THE INFLUENCE OF DIFFERENT PROCESSING PARAMETERS ON ESTIMATED VARIATION OF GPS POSITIONING

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Abstract

Throughout the past decades, GPS has been responsible much of the present infrastructure and thus being widely used covering from the diverse low cost and recreational uses to highly accurate and professional applications. Level of performance produce by this ingenious space-based positioning system however, is often a function of equipment suitability, measurement technique, error modelling and satellite signal geometry. Data processing complexity is also one of the main factors contributing to the erroneous and discrepancies in the GPS derived positions. This paper highlights the influence of three different processing parameters namely ephemeris type, elevation cut-off angle and earth rotation parameter towards the estimated variation of GPS positioning discrepancies. As each station experiencing low multipath conditions and excellent views of all available satellites, series of RINEX data retrieved from five continuously operating Malaysian Real-time Kinematic Network (MyRTKnet) stations namely JHJY, KLUG, JUML, MERU and LGKW were used. Based on the results, it is suggested that GPS derived positions varies with variation on the ephemeris type and elevation cut-off angle. As far as earth rotation parameter is concerned, it is totally ignorable in most practical cases as no variation detected throughout the observation period.

Keywords: batch processing interval, earth rotation parameter, elevation cut-off angle

1.0 INTRODUCTION

Over the past decades, Global Positioning System (GPS) has developed into an efficient yet practical satellite-based positioning and navigation tool capable of addressing various applications i.e. sports and recreational, surveying and mapping, petrology, open-pit mining, precision farming and earth deformation study. Emerging from the first launched of GPS satellite in 1978, practitioners are eventually reluctant to switch to a cumbersome theodolite or any other conventional terrestrial methods unless it is the only choice under some blocked environments. It is suggested that however, GPS measurements are subject to many error sources which degrade the obtainable accuracy of the final derived positions. Apart from the equipment suitability, measurement technique, error modelling and satellite signal geometry, pertinent data processing strategies are merely essential to compensate the risk of positioning displacement. Consequently, to attend the demand of high accuracy applications, a huge and rapid-growing quantity of GPS processing softwares have been developed and hence available in the market today. Ranging from commercial, in-house and scientific softwares, several studies have been made to investigate the performance of varying processing tools (Likhar et al. 2002; Even-Tzur et al. 2004). It is suggested that however, inadequate selection of processing parameters might significantly contributes to the variations in the discrepancies of the GPS measurement. To extend the understanding on the complexity of data processing, the impact of various processing parameters is assessed. This paper focuses on the statistical analysis of positioning variation induced by different orbit types, elevation cut-off angles and earth rotation parameter during GPS post-mission data processing. As the results of this study are not suggested to apply in all situations, the results shown in this paper however are likely to be duplicated and simulated under similar responsible conditions.

2.0 THE EXPERIMENT

2.1 Data Acquisition

Series of RINEX data retrieved from five continuously operating Malaysian Real-time Kinematic Network (MyRTKnet) were used throughout the study. These stations include Johor Jaya (JHJY), Kluang (KLUG), JUPEM Melaka (JUML), Meru (MERU) and Langkawi (LGKW). Apparently, all selected MyRTKnet stations are well-equipped with 5700 Trimble dual frequency geodetic GPS receivers with Zephyr Geodetic type antennas (Ali et al. 2006). The quality of the data retrieved from these stations is expectedly high, with each station experiencing low multipath conditions and excellent views of all available satellites. Located at the southern part of peninsula Malaysia, JHJY was selected as a reference in relative to other corresponding stations. The distance between KLUG, JUML, MERU and LGKW stations in relative to JHJY are respectively sufficient to signify short baseline (50 km), medium baseline (98 km), long baseline (193 km) and very long baseline (687 km). Furthermore, these baselines are best to represent the geographical condition in Peninsula Malaysia in which stretching from about 1 degree to 6 degrees latitudes. Figure 1 illustrates the geographical locations of the abovementioned stations.

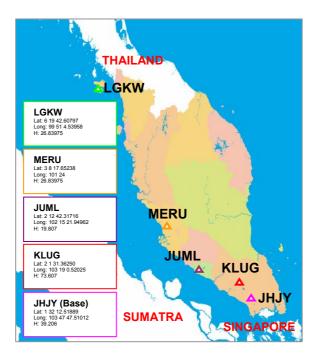


Figure 1: Distribution of Selected MyRTKnet Stations

2.2 Processing Methodology

A vendor-supplied processing software namely, TrimbleTM Geomatic Office v1.63 (TGO) was used throughout the study. TGO is an extensive commercial quality control tools that allows GPS baseline processing based on weighted ambiguity vector estimator. The main processing methodology follows the recommendations of GPS worldwide agencies i.e. International GNSS Service (IGS), EUREF permanent GPS Network (EPN) etc. Unlike in any other cases, the need of choosing the correct parameters is not as important as in this particular case. This is due to the fact that in the final solution, the influence of the choosen parameters will always be the same and thus the variation of the results will mainly depend on the variation of the "parameters of interest". Table 1 tabulates the summary of certain processing parameters used throughout the study.

Software	Trimble TM Geomatic Office v1.63
Processing Mode	Post-Processing (Relative Mode)
Solution	Ionosphere-free Double Difference Fixed
Tropospheric Model	Saastamoinen Model (Standard Atmosphere)
Orbit Type	IGS Final orbit, Broadcast orbit
Elevation Cut-off Angle	0°, 5°, 10°, 15°, 20°, 25°, 30°, 35°, 40°, 45°
Earth Tides	IGS Final Earth Rotation Parameter, No Parameter

Table 1: Summary of Processing Parameters

2.3 Data Analysis

As tabulated in Table 1, the so called "parameters of interest" includes ephemeris types, elevation cutoff angles and earth rotation parameter. To gauge the contribution of these processing parameters to variation in discrepancies of GPS derived positions, each case is then estimated and analyzed in term of 3D error based on Equation (1).

$$3DError = \left(\left(\Delta E\right)^2 + \left(\Delta N\right)^2 + \left(\Delta H\right)^2\right)^{\frac{1}{2}}$$
(1)

where:

3.0 RESULT

3.1 Ephemeris Data

Ephemeris data consists of broadcast orbits and precise orbits. Broadcast orbits is transmitted in the navigation message and uses Keplerian elements to signify the satellite orbits and incorporates additional terms to account the effects of the perturbing forces. While there are several disturbing forces that can be readily computed; others in particular the smaller forces, require detailed modelling and are still subject to further research (Leick 2005). Precise orbits file on the other hand consists of accurate positional data and clock corrections for each satellite for some specified period. Normally composed in an ASCII format (.sp3) or in binary counterpart (.ef18), precise orbits files can be downloaded from a variety of sources including IGS, U.S. National Geodetics Survey and Scripps Orbit and Permanent Array Center (SOPAC). There are three types of precise orbits namely ultra rapid orbits, rapid orbits and final orbits. The respective accuracy of these orbits (20 cm, 10 cm and 5 cm) have been determined by Hugentobler et al. (2001). To improve erroneous measurement caused by the orbital error (as well as other associated GPS errors), it is frequently recommended that precise orbits (or also known as post-mission orbits) need to be used in place of the broadcast orbits (Heroux & Kouba 2001). To investigate the positioning variation induced by different ephemeris data, 24 hours of RINEX daily data retrieved from JHJY-KLUG baseline were used. Given that other processing parameters were kept constant throughout five days of observation campaigns, results of the study are as shown in Figure 2.

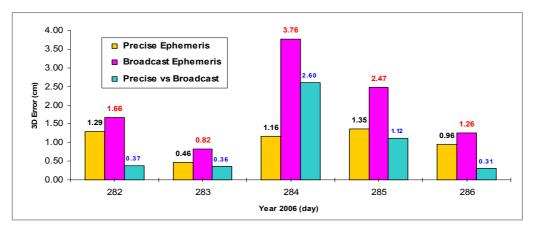


Figure 2: Computed 3D Error as a Function of Orbit Type

The result indicates that precise orbits tend to contribute better results in the computation of GPS derived positions in comparison to broadcast orbits. As opposed to broadcast orbits, on Day 284, the maximum variation from the use of the precise orbits is about 2.6 cm or 69 percents. Improvement on the order of about 0.4 cm or 44 percents is seen on the previous observation period (Day 283), followed by 1.1 cm or 45 percents, on Day 285. To further understand the variation of this parameter, computed 3D error was then calculated as a function of four different baselines; short baseline (JHJY-KLUG), medium baseline (JHJY-JUML), long baseline (JHJY-MERU) and very long baseline (JHJY-LGKW) given specific observation period. Figure 3 depicts the result of the study.

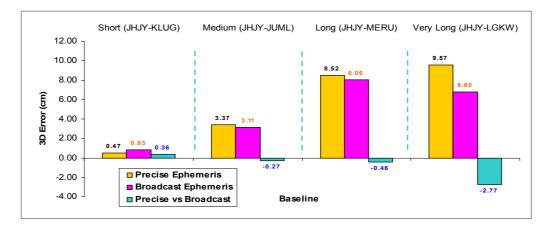


Figure 3: Computed 3D Error as a Function of Orbit Type and Baseline Length

Even so 3D measurement error increases with the increases distance over two reference stations; there are interesting correlations between the functionality of these orbit types and GPS baseline length. Short baseline (JHJY-KLUG) tends to evident the benefit of using the precise orbits. Longer baselines on the other hand considerably indicate better results of not using precise orbits. The variation of 3D measurement errors computed from both medium (JHJY-JUML) and long baselines (JHJY-MERU) range within decimal level of 0.3 cm to 0.5 cm. Similarly, although it is not that apparent for both aforementioned baselines, the largest and most noticeable variation of using broadcast orbits, is detected on the very long baseline (JHJY-LGKW). With a baseline length of about 687 km, positioning variation on the order of 2.8 cm or 29 percents is detected. Again using both precise and broadcast ephemeredes, Figure 4 (a, b and c) visualize the variations in term of Easting, Northing and Height components of GPS derived positions over similar observation period.

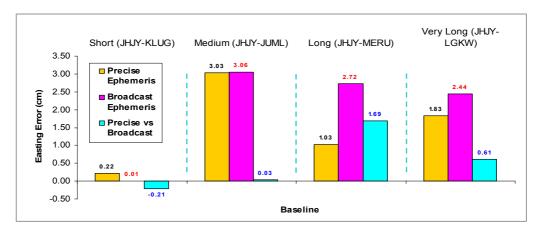


Figure 4 (a): Computed Easting Error as a Function of Orbit Type and Baseline Length

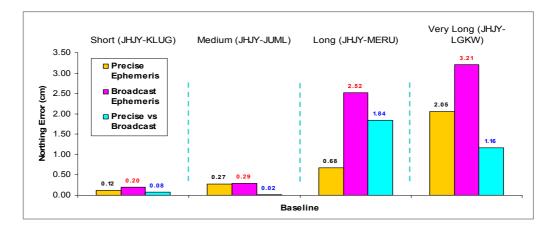


Figure 4 (b): Computed Northing Error as a Function of Orbit Type and Baseline Length

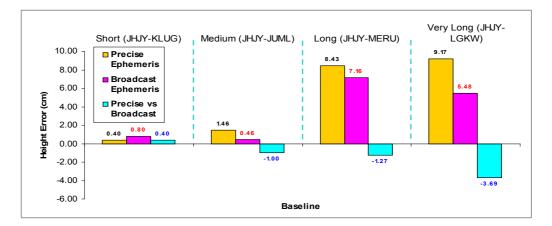


Figure 4 (c): Computed Height Error as a Function of Orbit Type and Baseline Length

3.2 Elevation Cut-off Angles

Another parameter to be taken into account in data processing is the elevation cut-off angle. Often denoted to elevation mask angle, it is a setting for the tracked satellites angle in relative to the receiver's antenna. As signal perturbation is most likely within unfavourable observation surrounding (i.e. reflective surface, blocked structure), it is suggested that high elevation cut-off angle removes noisy data and therefore improves baseline statistics. Furthermore, signals transmitted from low elevation satellites need to propagate through longer atmosphere and are subject to ionospheric and tropospheric effects. Multipath effect is also much more pronounced among low elevation satellites than signals that are directly overhead. Exclusion of low orbiting satellites is therefore needed to ensure the performance of GPS measurement. To further comprehend the effect of different elevation cut-off angle on computed 3D error, a study was conducted based on three respective periods (Day 283 to 285). Different elevation cut-off angles were set ranging from 0 degree to 45 degrees. With a constant batch processing interval from 8 am to 9 am at JHJY-KLUG baseline, estimated variation of GPS positioning induce by different elevation cut-off angle are as shown in Figure 5.

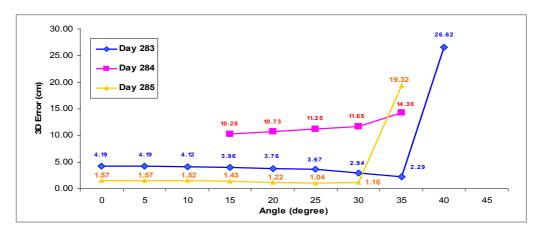


Figure 5: Computed 3D Error as a Function of Elevation Cut-off Angle

For this particular result, the distributed 3D error varies as a function of elevation cut-off angles. Although no fixed solution can be made at 45 degrees angle during all observation period, on Day 283, an increase in elevation cut-off angle gradually improves the computed 3D errors. The best computed GPS position is detected at the 35 degrees angle with 2.3 cm accuracy, or 45 percents of improvement in comparison to 4.2 cm at 0 degree angle on the same day. Nevertheless, once the elevation reaches 40 degrees angle, an excessive 3D positioning displacement of 26.6 cm is detected and hence should be avoided. An increase in elevation cut-off angle on Day 284 on the other hand, decreases the accuracy level obtained from the GPS data processing. Generally produced less accurate measurement in comparison to Day 283, the best computed GPS position is detected at the lowest possible angle of 15 degrees, given that only float solution can be produced for the first and last 10 degrees. Although on Day 285, similar pattern to Day 283 is detected (in which an increase in elevation cut-off angle gradually improves the measurement accuracy), for this particular case however, sudden fluctuation is detected at 35 degrees angle. To extend the understanding on the variation of this parameter, computed 3D errors were then calculated as a function of both elevation cut-off angle and observation period. Analyses were made based on hourly interval on Day 283 from 8 am to 4 pm. Figure 6 depicts the result of the study.

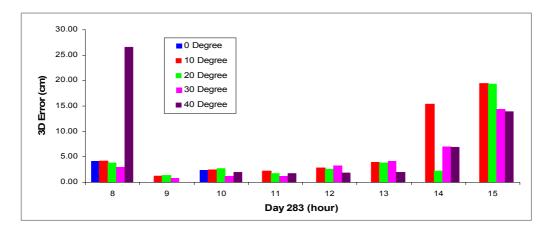


Figure 6: Computed 3D Error as a Function of Elevation Cut-off Angle and Observation Period

As mentioned earlier, the need for elevation cut-off angle is to characterize the level or viewpoint from where visible satellites in reference to the receiver will be included. Due to the relative motion of the orbiting satellites, inappropriate selection of elevation cut-off angle induces discrepancies in GPS derived positions. Before or even after (in a case where no solution available at 0 degree) reaching to sudden fluctuation, float or even no wave solution due to insufficient amount of satellites (minimum of four satellites is needed to obtain one 3D position), it is suggested that increase in elevation cut-off angle gradually improves the measurement accuracy. In most cases, it is also noted that within 20 to 30 degrees angle, the 3D errors tends to be at the smallest amount. This might be due to the fact that only good and 'healthy' satellites are included within this observing angle. To justify the significant of elevation cut-off angle setup with variation in satellite visibility, Figure 7 illustrates the relative motion of the orbiting satellites over similar observation period (Day 283) at JHJY-KLUG baseline.

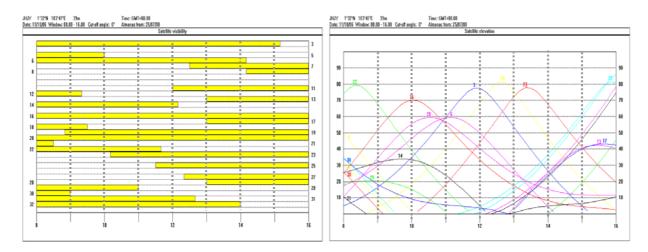


Figure 7: Multi Station Analysis on Satellite Visibility and Elevation

Here, strong correlation between elevation cut-off angle setup (Figure 6) and satellite condition (Figure 7) over similar observation period is detected. Provided that enough satellites with high elevation are available at that particular period, it is suggested to use high elevation cut-off angle in substitute of low elevation angle. In a situation where fewer amounts of satellites are available, the inclusion of low elevation satellites however, should be considered. As signals transmitted from low elevation satellite are more susceptible to multipath and atmospheric disturbance, appropriate site selection with multipath-resistant antenna is therefore recommended.

3.3 Earth Rotation Parameter

Another parameters used in data processing is the earth rotation parameter (ERP). ERP generally consists of pole position and UT1-UTC along with the conventions for sidereal time, precession and nutation. Normally provided in three types of grouped parameters (i.e. polar motion, polar motion rates and length of day), ERP can be estimated either using VLBI or continuous GPS monitoring. ERP file can be downloaded via IGS website at no direct charge. Detail descriptions of ERP are as discussed in Hinnov (2004). ERP in general is used to mitigate the effect of pole tide (details on pole tide mathematical expression is as discussed in IERS (1996)). Schuler (2001) asserts that site deformation due to polar motion can reach a maximum radial displacement of about 2.5 cm. Nevertheless, this so called effect is noted to be highly dependent on the site location. In order to investigate the significant of using ERP towards variation in GPS positioning accuracy within the area of selected reference stations, analyses were made based on JHJY-LGKW baseline processed on five corresponding observation period. For the purpose of study, final ERP were used in which can be obtained at about 13 days latency. Results of the study are as shown in Figure 8.

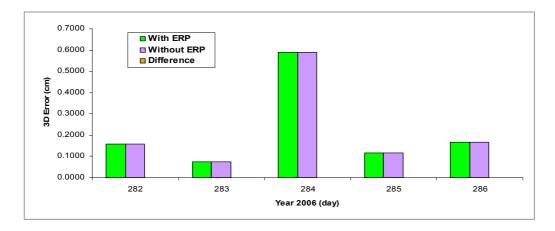


Figure 8: Computed 3D Error as a Function of Earth Rotation Parameter

Apparently, based on this particular case, the use of ERP however had no effect on the GPS derived positions. 3D errors induced either by using ERP or without using ERP throughout the observation campaign (Day 282 – Day 286) are entirely similar. The use of ERP during data processing is therefore can be neglected for most practical cases including local high accuracy GPS surveying.

4.0 CONCLUSION

This aim of this paper is to investigate GPS positioning variation caused by differences in orbit types, elevation cut-off angles and earth rotation parameters. Based on the analyses (see Figure 2 – Figure 8) conducted within series of RINEX data retrieved from selected continuously operating MyRTKnet stations, several conclusions are made:

- 1. GPS derived positions varies with variation on the orbit types. Furthermore, 3D measurement error caused by different orbit types increases with the increases distance over two reference stations. As the estimated variation evidently depends on the baseline length, the largest and most noticeable variation on the order of 2.8 cm is detected on the very long baseline of about 687 km.
- 2. GPS derived positions varies with variation on the elevation cut-off angle. Depending on the relative motion of the orbiting satellites, it is suggested that increases in elevation cut-off angles gradually improve the measurement accuracy. In most cases, it is also noted that within 20 to 30 degrees angle, the 3D errors tends to be at the smallest amount. In a situation with

low satellites visibility, the inclusion of low elevation satellites however, should be considered.

3. As far as earth rotation parameter is concerned, it is totally ignorable in most practical cases as no variation detected on the accuracy of GPS positioning throughout the observation period.

The results of this study are not suggested to apply in all situations. To effectively quantify the significant of these parameters in data processing, the use of massive amount of data is highly recommended. The results shown in this paper however are likely to be duplicated and simulated under similar responsible conditions.

5.0 ACKNOWLEDGMENT

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