

NUMERICAL ANALYSIS OF DETONATION STABILITY IN A ROTATING  
DETONATION ENGINE FUELLED WITH BIOGAS AND HYDROGEN

MOHAMMAD NURIZAT RAHMAN

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## DEDICATION

This thesis is dedicated to my mother, my greatest supporter and pillar in my life, who taught me that the finest kind of knowledge to have is knowledge gained for the benefit of others.

To my father, who ingrained in me the belief that hard work pays off.

To Amaleen, who has been a strong emotional supporter throughout my doctoral study and has also assisted me in becoming the best version of myself.

To Along, who was the first to open my eyes to the difficulties of living independently and inspire me to become the strong person I am today.

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## ABSTRACT

The novel rotating detonation engine (RDE) fuelled with biogas offers a significant contribution to the application of combustion engines powered by renewable-based fuels. However, the potential of a biogas-fuelled RDE has never been properly examined in terms of key operating parameters such as ignition intensity, equivalence ratio, and total mass flow rate (MFR). Hence, the primary research goal for the current numerical study was to examine the stability of continuous rotating detonation waves (CRDW) in RDEs powered by hydrogen and biogas on the basis of the aforementioned operating parameters. The numerical model of CRDW was first established to represent the CRDW stability. Following that, the modified one-step chemistry for biogas detonation was developed and merged with the validated CRDW numerical model. The impact of the above-mentioned critical operating parameters on CRDW stability in the biogas-fuelled RDE was explored using the validated CRDW numerical model, which was merged with the modified one-step chemistry for biogas detonation. The CRDW numerical model revealed that the predicted CRDW pressure was within 10% of the experimental data. The one-step model was compared to experimental data and the detailed chemistry data, revealing 15.75% and 8.29% discrepancies in biogas detonation velocities. The result is that in a fuel-lean non-premixed environment at fixed ignition intensities, the biogas-fuelled RDE outperformed the hydrogen counterpart in terms of detonation stability, with the predicted time to achieve a stable one-wave CRDW in the former RDE being 1327 microseconds shorter than that of the latter RDE. However, the former RDE fell short in detonation sustainability, as predicted by the wave longevity. After 0.0146 seconds from the one-wave emergence, the CRDW was extinguished in the former RDE, while the CRDW pressure was only decreased by 1.52% in the latter RDE. The fundamental explanation for this was that biogas, which has lower diffusivity and reactivity than hydrogen, created an imbalance in counter-rotating waves, resulting in a faster CRDW mode transition than hydrogen. Multiple collisions of counter-rotating waves have been discovered to be the primary mechanism in the CRDW stabilization process. There was a balance between gaining and losing energy for counter-rotating waves, culminating in a CRDW mode transition or CRDW extinguishment. The enhanced ignition intensity, equivalence ratio, and MFR produced the expected increase in CRDW intensity in the biogas-fuelled RDE. Enhancing these parameters aided in boosting the detonability of the biogas-air mixture. Quasi one-wave CRDW was observed from the start of ignition in all parametric cases, showing that the state of chaotic detonation instability was hardly occurred using biogas. To conclude, the current study discovered that the CRDW from the biogas-fuelled RDE has a more comprehensive operating stability range than the hydrogen-fuelled counterpart. Still, the rapid biogas detonation decay highlights the necessity for an enhanced mixing rate to preserve detonation continuity. The assessment of CRDW instabilities in the current study is pivotal for ensuring that these instabilities are effectively regulated and taken into account during the establishment of a RDE powered by biogas. The findings will also spearhead further research into parameters that could sustain CRDWs in the future working prototype of biogas-fuelled RDE.

## ABSTRAK

Enjin ledakan berputar (RDE) digerakkan dengan biogas memberikan sumbangan kepada penciptaan enjin pembakaran yang dikuasakan oleh bahan bakar berasaskan tenaga boleh diperbaharui. Tetapi, potensi biogas RDE tidak pernah disiasat secara menyeluruh berdasarkan kritikal parameter operasi seperti intensiti pencucuhan, nisbah kesetaraan, dan kadar aliran jisim total (MFR). Jadi, tujuan utama kajian berangka ini adalah untuk meramal kestabilan perambatan gelombang ledakan berputar berterusan (CRDW) di RDE yang dibekalkan dengan hidrogen dan biogas berdasarkan impak parameter di atas. Untuk mewakili kestabilan CRDW, model berangka CRDW telah dibina. Satu model kimia satu langkah diubah suai telah dibina untuk peledakan biogas dan telah digabungkan dengan model berangka CRDW. Penilaian parametrik terhadap parameter ini telah dijalankan melalui model berangka CRDW yang telah disahkan bersama-sama model kimia satu langkah diubah suai untuk peledakan biogas. Ramalan tekanan CRDW telah berjaya disahkan dengan eksperimen dengan ralat kurang dari 10%. Model kimia satu langkah ini juga telah dibandingkan dengan eksperimen dan mekanisme terperinici, yang mana perbezaan masing-masing adalah 15.75% dan 8.29% terhadap halaju peledakan telah diperhatikan. Keputusannya, di persekitaran kurang bahan bakar dan tanpa pracampuran dengan nilai pencucuhan sama, biogas RDE melebihi hidrogen RDE dari segi kestabilan peledakan yang mana ramalan masa untuk mencapai satu gelombang stabil di biogas RDE adalah 1327 mikrosaat lebih pendek dari hidrogen RDE. Namun, biogas RDE kurang dari segi keberlanjutan peledakan seperti yang diramalkan melalui kesinambungan gelombang. Selepas 0.0146 saat dari kemunculan satu gelombang stabil, RDW telah terpadam di biogas RDE tetapi hanya berlaku 1.52% pengurangan tekanan CRDW di hidrogen RDE. Asas penjelasan kepada perkara ini ialah biogas, yang mempunyai difusiviti dan kereaktifan yang lebih rendah daripada hidrogen, mewujudkan ketidakseimbangan dalam gelombang putaran balas, dan menghasilkan peralihan mod CRDW yang lebih cepat daripada hidrogen. Perlanggaran berbilang gelombang putaran balas telah ditemui sebagai mekanisme utama dalam proses penstabilan CRDW. Terdapat keseimbangan antara memperoleh dan kehilangan tenaga untuk gelombang berputar balas, yang memuncak dalam peralihan mod CRDW atau pepadaman CRDW. Penggunaan intensiti penyalaan, nisbah kesetaraan dan MFR yang tinggi dijangka meningkatkan intensiti CRDW kerana peningkatan parameter ini akan menghasilkan keupayaan ledakan yang lebih tinggi untuk campuran biogas-udara. Kuasi satu gelombang CRDW berlaku dari awal pencucuhan di kesemua kes parametrik, menunjukkan ketidakstabilan ledakan sukar untuk dicapai melalui biogas. Kesimpulannya, CRDW dalam biogas RDE mempunyai julat kestabilan yang lebih luas daripada hidrogen RDE. Tetapi, pereputan ledakan biogas yang cepat menyerlahkan keperluan untuk meningkatkan kadar pencampuran untuk mengekalkan kesinambungan ledakan. Penilaian ketidakstabilan CRDW dalam kajian semasa adalah penting untuk memastikan ketidakstabilan ini dikawal dengan berkesan dan diambil kira semasa pembinaan biogas RDE. Penemuan ini juga akan menjadi batu loncatan untuk penelitian lebih lanjut parameter yang dapat memanjangkan keberlanjutan CRDW dalam biogas RDE.

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## LIST OF ABBREVIATIONS

RDE	-	Rotating detonation engine
CRDW	-	Continuous rotating detonation wave
MFR	-	Total mass flow rate
SST	-	Shear stress transport
PDE	-	Pulse detonation engine
DT	-	Detonation tube
RANS	-	Reynolds-averaged Navier–Stokes
DDT	-	Deflagration to detonation transition
CJ	-	Chapman-Jouguet
ZND	-	Zeldovich-von Neumann-Döring
LHV	-	Lower heating value
LIH	-	Lavrentyev Institute of Hydrodynamics
TRL	-	Technology Readiness Level
HiREF	-	High Speed Reacting Flow Research Laboratory
UTM	-	Universiti Teknologi Malaysia
NASA	-	National Aeronautics and Space Administration
ROP	-	Rate of development
DAQ	-	Data acquisition
DNS	-	Direct Numerical Simulation
LES	-	Large Eddy Simulation
CFD	-	Computational fluid dynamics
3D	-	Three-dimensional
2D	-	Two-dimensional
NS	-	Navier–Stokes
RNG	-	Re-normalisation group
RSM	-	Reynolds stress equation model
CEA	-	Chemical equilibrium with applications
EDC	-	Eddy Dissipation Concept
AUSM	-	Advection Upstream Splitting Method
PT	-	Pressure transducer

PT1	-	Pressure transducer 1
PT2	-	Pressure transducer 2
CPU	-	Central processing unit
a.c.	-	Alternating current
d.c.	-	Direct current
RV	-	Reducing valve
MV	-	Magnetic valve
CV	-	Check valve
SP	-	Spark plug
M	-	Ignition modular
C	-	Ignition coil
PC	-	Personal computer
RT2	-	Rectifier
D2	-	Diode
C2	-	Capacitor arrangement
RL2	-	Relay
SWACER	-	Shock wave amplification by coherent energy release
ER0.6	-	Equivalence ratio of 0.6
ER0.7	-	Equivalence ratio of 0.7
ER0.8	-	Equivalence ratio of 0.8
ER0.9	-	Equivalence ratio of 0.9
ER1.0	-	Equivalence ratio of 1.0
MF0.16	-	Total mass flow rate of 0.16
MF0.26	-	Total mass flow rate of 0.26
MF0.36	-	Total mass flow rate of 0.36
MF0.46	-	Total mass flow rate of 0.46
MF0.56	-	Total mass flow rate of 0.56



## LIST OF SYMBOLS

$\phi$	-	Equivalence ratio
$N_2$	-	Nitrogen
$CH_4$	-	Methane
$CO_2$	-	Carbon dioxide
$OH$	-	Hydroxyl radical
$H$	-	Hydrogen element
$O$	-	Oxygen element
$T$	-	Temperature
$u$	-	Velocity vector component in the $x$ -direction
$v$	-	Velocity vector component in the $y$ -direction
$w$	-	Velocity vector component in the $z$ -direction
$\rho$	-	Density
$P$	-	Pressure
$M$	-	Mach number
$\dot{m}/A$	-	Mass flux
$Q$	-	Heat addition
$R$	-	Specific gas constant
$c_p$	-	Specific heat capacity (constant pressure)
$\gamma$	-	Specific heat ratio
$A$	-	Area
$\lambda$	-	Reaction progress variable
$h$	-	Specific enthalpy
$c$	-	Speed of sound
$E_a$	-	Activation energy
$\Delta$	-	Change in a quantity
$\emptyset$	-	Universalization of any variable
$\sum \dot{Q}$	-	Net rate of heat added to the control volume
$\sum \dot{W}$	-	Net rate of work done on the control volume

$\dot{Q}_s \Delta V$	-	Rate of heat added or removed by heat source on the control volume
$\mu$	-	One of the proportionality constants related to viscosity
$\lambda_v$	-	One of the proportionality constants related to viscosity
$R_{ij}$	-	Reynold stresses
$k$	-	Turbulent kinetic energy
$\varepsilon$	-	Rate of dissipation of turbulent kinetic energy
$\omega$	-	Specific rate of dissipation
$J_i$	-	Diffusion flux of species $i$
$R_i$	-	Net rate of production of species $i$ by chemical reaction
$S_i$	-	Rate of creation by addition from the dispersed phase plus any user-defined sources
$N_R$	-	Reaction number
$M_{w,i}$	-	Molecular weight of species $i$
$\hat{R}_{i,r}$	-	Arrhenius molar rate of creation/destruction of species $i$ in reaction $r$
$N$	-	Number of species involves in the reaction
$v''_{i,r}$	-	Stoichiometric coefficient for product $i$ in reaction $r$
$v'_{i,r}$	-	Stoichiometric coefficient for reactant $i$ in reaction $r$
$k_{f,r}$	-	Forward rate constant of reaction $r$
$k_{b,r}$	-	Backward rate constant of reaction $r$
$C_{j,r}$	-	Molar concentration of species $j$ in reaction $r$
$\eta'_{j,r}$	-	One of the rate exponents of species $j$ in reaction $r$
$\eta''_{j,r}$	-	One of the rate exponents of species $j$ in reaction $r$
$t$	-	Time
$Y_i^*$	-	Fine-scale species mass fraction after reacting over the time scale, $\tau^*$ .
$\zeta^*$	-	Length fraction of the fine scales
$C_\xi$	-	Volume fraction constant
$C_\tau$	-	Time scale constant
$\Delta t_1$	-	Non-reacting mixing phase (filling stage)
$\Delta t_2$	-	Non-reacting mixing phase (filling stage)

$\Delta t_3$	-	Reacting flow phase (combustion period)
$T_m$	-	Flow rate temperature
$P_i$	-	Ignition pressure
$T_i$	-	Ignition temperature
$V_i$	-	Ignition velocity
$P_{CJ}$	-	CJ ignition pressure
$T_{CJ}$	-	CJ ignition temperature
$V_{CJ}$	-	CJ ignition velocity
$t_o$	-	Time interval exceeding the time scales of the slowest variations from the largest eddies
$\bar{u}$	-	Mean velocity
$G_k$	-	Generation of $k$
$G_\omega$	-	Generation of $\omega$
$\Gamma_k$	-	Effective diffusivity of $k$
$\Gamma_\omega$	-	Effective diffusivity of $\omega$
$Y_k$	-	Dissipation of $k$
$Y_\omega$	-	Dissipation of $\omega$
$k_r$	-	Generic rate constant
$A_r$	-	Pre-exponential factor
$n$	-	Temperature exponent
$x$	-	Upstream annulus offset length
$\alpha_1$	-	Upstream annulus divergence angle
A1	-	Number of fuel inlets
A2	-	Distance between fuel and oxidizer inlets
A3	-	Fuel inlet diameter
A4	-	Annulus width
A5	-	Inner annulus radius
A6	-	Outer annulus radius
A7	-	RDE body length

# CHAPTER 1

## INTRODUCTION

### 1.1 Background

The rotating detonation engine (RDE) is an ingenious energy conversion system in such it promises remarkable reduction in fuel consumption and emissions [1]. Utilisation of detonation mode is the pinnacle feature of a RDE as it has the capability to liberate energy much quicker than deflagration combustion mode which take place in the conventional gas turbine system [2]. To further enlighten the understanding of RDE working operation, two major modes of combustion needs to be described properly. Detonation and deflagration are referred as the supersonic and the subsonic combustion processes respectively. The detonation mode triggered by the initiation of shock wave propagates through the flammable mixture with a supersonic speed. The characteristics of detonation including high thermal efficiency, rapid release of energy, and shockwave across the combustion region, and thus, makes it favourable in power and propulsion studies.

A RDE not only has all of the advantages of detonation, but also has the advantage of the compact and simple structure. With pressure gain attribute in detonation phenomenon, it can minimise the pressure that is required to compress the incoming air in the gas turbine system, thus, contribute to less compressor stages as compared to the conventional gas turbine systems. The current gas turbine systems utilise deflagration mode to burn the reactant mixture. The deflagration mode requires multi-stages compressors to elevate the pressure prior to being burned in a combustor. Therefore, the implementation of a RDE can simplify the engine structure, reduce the engine weight, and decrease the fuel consumption [2]. A RDE can provides nearly constant thrust using continuous rotating detonation waves (CRDWs) [2]. The concept of CRDW in a RDE makes it a better energy conversion system than the prior

detonation-based energy conversion system; the pulse detonation engine (PDE) which used intermittent detonations [3-4].

The concept of a non-fossil fuel economy has now become the target of many regions around the world. In the spirit of developing a novel combustion engine powered by renewable-based fuels, the already novel RDE, when combined with the potential to operate with renewable-based fuels such as biogas and green hydrogen, will create a major contribution to the creation of a novel combustion engine powered by renewable-based fuels. With its high potential for incorporation into a gas turbine system by replacing the traditional deflagration combustion mode, the future of a RDE powered by renewable-based fuels is bright. As a result, the RDE system will bring an innovative revolution to existing power generation, aviation, and aerospace propulsion systems, as well as contribute to the reduction of global warming [5]. To accomplish this aim, the creation of a dependable and functional RDE system powered by renewable-based fuels such as biogas and green hydrogen is needed.

## **1.2 Problem Statement**

In experiments to detect CRDWs in RDEs, pressure transducer and ion probe readings are commonly used [2]. Despite the vast amount of experimental data that can be taken from these sensors and exploited to discover different CRDW peculiarities [2], there is still a lack of information as there is a dearth of an overall picture of the CRDW process due to experimental setup constraints. Having said that, the point-based readings from the pressure transducer and the ion probe are insufficient to provide complete insights into CRDW dynamics. The constraints of experimental methodologies for displaying CRDW dynamics hampered in-depth understanding of CRDW phenomena, particularly CRDW stabilization and sustainability. When compared to experiments, numerical analysis can greatly improve and simplify CRDW visualisation. However, the lack of reliable numerical models that can improve understanding of CRDW stabilization will result in major instabilities not being properly identified during the CRDW stabilization stage, and the "indiscernible" difficulties not being appropriately addressed in order to construct a workable RDE,

particularly one powered by biogas or hydrogen. Therefore, in order to construct a feasible RDE system, a credible numerical model that can improve understanding of CRDW stabilization and sustainability is required, which will eventually be employed for simulation-based design processes. A reliable numerical model can also be used to optimise RDE injection structures in order to improve mixing processes in the RDE annulus, which appears to be lacking in experimental studies that have leveraged the advantage of the RDE numerical model to improve mixing processes.

The chemistry model is one of the most critical factors to take into account when constructing a reliable numerical model that can accurately predict the CRDW dynamics in RDEs. The development of an appropriate chemistry model for biogas detonation is critical for predicting CRDW dynamics in a biogas-fuelled RDE. The validation feedback loop at varied constants in a chemistry is often executed until the simulation results agree well with the experimental data. While there is almost no experimental study on the use of biogas in RDEs, there are a few experimental publications, such as Wahid [6], Ghazali [7], and Elhawary [8], that have experimentally tested biogas detonation on detonation tube-based engines. The fitting of a chemical model to simulate biogas detonation is critical because the comparatively low reactivity and diffusivity of biogas [9] could lead to diverse reacting flow dynamics in RDEs. If the chemistry model is not calibrated to represent biogas detonation, significant CRDW phenomena such as CRDW stabilization [10-11], unstable CRDWs [12], and detonation extinguishment [13-19] cannot be accurately captured in a biogas-fuelled RDE.

While there have been a number of detonation studies that have employed one-step and two-step Arrhenius chemistry models to account for both the accuracy and efficiency of detonation modelling [20-31], all of these studies addressed hydrogen detonation and hence used a hydrogen-air/oxygen chemistry model. Therefore, the published one-step chemical models have not been validated or tuned for biogas detonation modelling. As a result, there is some doubt about the credibility of the published one-step Arrhenius chemistry model to well predict the detonation behaviour in biogas-fuelled RDEs, as there is a high possibility that the published one-step models will produce ignition delay times that are a few orders of magnitude different than the ignition delay times from biogas detonation [32]. To appropriately

model the CRDW from the biogas-powered RDE, the constant in the one-step Arrhenius chemistry model should be adjusted via the validation feedback loop to reflect the biogas detonation experimental data.

In terms of RDE systems, the construction of a fuel-lean non-premixed RDE is appealing in terms of safety and fuel consumption reduction. Nonetheless, in terms of CRDW dynamics, previous studies have found a number of serious concerns with the CRDW sustainability in a non-premixed RDE, the most notable of which being CRDW stability and longevity. Because hydrogen is the most commonly investigated fuel in the realm of RDE research, the majority of these issues were discovered in hydrogen-fuelled RDEs. Several research have already discovered the transient stabilization period for achieving a stable CRDW. Both Ma [10] and Liu [11] have scrutinized the events that lead to the formation of a stable CRDW. However, both of these references used premixed reactants. There are scenarios in non-premixed cases where it ends up producing unstable CRDWs [12] or even worse, the detonation extinguishment happened [13-18]. The phenomenon of detonation extinguishment in a non-premixed RDE is much more common in which a detonation failed to be created in the RDE annulus, ends up becoming a normal deflagration or no combustion at all. In a leaner equivalence ratio, the severity of CRDW instabilities was observed to be worsened [19]. All of these findings indicate that there is still a significant knowledge gap on the CRDW stabilization process in a fuel-lean non-premixed RDE, with one of the key reasons being the limits in experimental methodologies for visualising CRDWs, which inhibited in-depth understanding of CRDW dynamics. Characterization of instabilities during the CRDW stabilization process in a fuel-lean non-premixed RDE is critical to ensure that instabilities are adequately regulated and taken into consideration during the development of a fuel-lean non-premixed RDE, especially one powered by biogas or hydrogen.

The biogas-fuelled detonation engine, however, is not a new concept. In fact, studies have previously been conducted to evaluate biogas detonation with an emphasis on practical use of the detonation engine [6-8]. Nonetheless, all of these studies have concentrated on detonation tube-based applications, such as the PDE, which have a lower detonation frequency than a more novel detonation engine, such

as the RDE. Hence, it is difficult to achieve a high detonation frequency in a PDE powered by low reactivity fuels such as pure biogas [6-8], making it unsuitable for practical usage. That being said, the CRDW, a quasi-continuous detonation with constant thrust in a RDE, could be the suitable combustion mechanism to allow the use of low reactivity fuels like biogas in detonation-based engines. However, hardly no study has been conducted on the use of biogas for RDEs. To investigate the potential of CRDWs in a biogas-fuelled RDE, a parametric assessment of critical operating parameters must be carried out. The equivalence ratio [2, 17, 19], mass flow rate [16, 24,] and ignition intensity [26, 76] are all significant operating parameters in RDE systems that have been examined previously. Despite this, there have been almost no studies that have investigated the effects of these parameters on CRDW stability in a biogas-fuelled RDE. As a result, the potential of a biogas-fuelled RDE has never been thoroughly investigated, leaving this type of engine with an unknown potential.

### **1.3 Research Questions**

The research questions are:

- (a) In comparison to experimental data, how accurate is the numerical model based on the Reynolds-averaged Navier–Stokes (RANS) equation and the one-step Arrhenius chemistry model in predicting the CRDW behaviour in RDEs?
- (b) Is the above-mentioned numerical model capable of reasonably predicting the transitory detonation stabilization phase in RDEs?
- (c) Is the modified one-step chemistry model capable of reasonably predicting the occurrence of biogas detonation?
- (d) Does the biogas-fuelled RDE used in the present study have a transitory detonation stabilization phase? If so, how does the transitory detonation stabilization phase from the biogas-fuelled RDE vary from the hydrogen-fuelled RDE?



- (e) What is the expected intensity and stability of CRDWs in fuel-lean non-premixed biogas- and hydrogen-fuelled RDEs?
- (f) What is the expected intensity and stability of CRDWs in the biogas-fuelled RDE at different ignition intensities, equivalence ratios, and mass flow rates?

#### **1.4 Research Objectives**

The research objectives are:

- (a) To establish the validated numerical model of CRDW based on the detonation experimental data.
- (b) To establish a comprehensive comparative assessment of CRDW behaviour from fuel-lean non-premixed biogas- and hydrogen-fuelled RDEs.
- (c) To predict the impact of ignition intensity, equivalence ratio, and mass flow rate towards the CRDW stability in the biogas-fuelled RDE.

#### **1.5 Scopes of Research**

The scopes of the present research are as follows:

- (a) The CRDW simulation in the hydrogen-fuelled RDE is limited to a single case as the baseline. The subsequent parametric studies are only for the biogas-fuelled RDE.
- (b) For objective (a), both experimental and numerical methods are used to establish the validated numerical model of CRDW.
- (c) For objectives (b) and (c), numerical methods are employed using the established validated numerical model in objective (a).

- (d) To capture the mixing process and CRDW behaviour, both non-reacting and reacting flows are transiently modelled.
- (e) The only geometry covered in the current analysis is the RDE body, which plays a significant role in the flow dynamics behaviour within the RDE system.
- (f) The current research focuses solely on the intensity and stability of CRDW in the biogas-fuelled RDE in terms of the predicted pressure front, as this is the most common parameter examined in experimental studies via pressure transducer readings. To back up the pressure front findings, the predicted heat of reaction rate is also analysed.
- (g) The predicted CRDW stability in the biogas-fuelled RDE is investigated via the decay of CRDW pressure and heat of reaction over time, with simulation time limited to the 12<sup>th</sup> cycle of CRDW propagation to balance simulation accuracy and efficiency.
- (h) The parameters studied in the biogas-fuelled RDE parametric studies are only the ignition intensity, equivalence ratio, and total mass flow rate.
- (i) Since it is the most typical biogas composition in the literature, this analysis only covers biogas with a composition of 65% methane (CH<sub>4</sub>) and 35% carbon dioxide (CO<sub>2</sub>). The impact of varying biogas composition on the CRDW stability is left for future research.

## REFERENCES

1. Sun, J., Zhou, J., Liu, S., Lin, Z., & Lin, W. (2019). Numerical investigation of a non-premixed hollow rotating detonation engine. *International Journal of Hydrogen Energy*, 44(31), 17084–17094.
2. Zhou, S., Ma, H., Ma, Y., Zhou, C., Liu, D., & Li, S. (2018). Experimental study on a rotating detonation combustor with an axial-flow turbine. *Acta Astronautica*, 151, 7–14.
3. Bengoechea, S., Gray, J. A., Reiss, J., Moeck, J. P., Paschereit, O. C., & Sesterhenn, J. (2018). Detonation initiation in pipes with a single obstacle for mixtures of hydrogen and oxygen-enriched air. *Combustion and Flame*, 198, 290–304.
4. Fujii, J., Kumazawa, Y., Matsuo, A., Nakagami, S., Matsuoka, K., & Kasahara, J. (2017). Numerical investigation on detonation velocity in rotating detonation engine chamber. *Proceedings of the Combustion Institute*, 36(2), 2665–2672.
5. Zhou, R., Wu, D., & Wang, J. (2016). Progress of continuously rotating detonation engines. *Chinese Journal of Aeronautics*, 29(1), 15–29.
6. Wahid, M. A., & Ujir, H. (2012). Reacting shock waves characteristics for biogas compared to other gaseous fuel.
7. Ghazali, A. D. (2015). Experimental Analysis of Pulse Detonation Engine Fuelled by Biogas Mixtures (Unpublished master's thesis). Universiti Teknologi Malaysia, Johor, Malaysia.
8. Elhawary, S., Saat, A., Wahid, M. A., & Ghazali, A. D. (2020). Experimental study of using biogas in pulse detonation engine with hydrogen enrichment. *International Journal of Hydrogen Energy*, 45(30), 15414-15424.
9. Suhaimi, M. S. (2020). Biogas Combustion Characteristics Under Varying Carbon Dioxide Dilution and Hydrogen Enrichment (Unpublished PHD's thesis). Universiti Teknologi Malaysia, Johor, Malaysia.
10. Ma, Z., Zhang, S., Luan, M., Yao, S., Xia, Z., & Wang, J. (2018). Experimental research on ignition, quenching, reinitiation and the stabilization process in rotating detonation engine. *International Journal of Hydrogen Energy*, 43(39), 18521–18529.

11. Liu, Y., Wang, Y., Li, Y., Li, Y., & Wang, J. (2015). Spectral analysis and self-adjusting mechanism for oscillation phenomenon in hydrogen-oxygen continuously rotating detonation engine. *Chinese Journal of Aeronautics*, 28(3), 669–675.
12. Wang, Y., Le, J., Wang, C., & Zheng, Y. (2018). A non-premixed rotating detonation engine using ethylene and air. *Applied Thermal Engineering*, 137, 749–757.
13. Peng, H., Liu, W., Liu, S., & Zhang, H. (2018). Experimental investigations on ethylene-air Continuous Rotating Detonation wave in the hollow chamber with Laval nozzle. *Acta Astronautica*, 151, 137–145.
14. Anand, V., George, A. S., Driscoll, R., & Gutmark, E. (2016). Investigation of rotating detonation combustor operation with H<sub>2</sub>-Air mixtures. *International Journal of Hydrogen Energy*, 41(2), 1281–1292.
15. Zhou, S., Ma, H., Chen, S., Zhong, Y., & Zhou, C. (2019). Experimental investigation on propagation characteristics of rotating detonation wave with a hydrogen-ethylene-acetylene fuel. *Acta Astronautica*, 157, 310–320.
16. Sun, J., Zhou, J., Liu, S., Lin, Z., & Lin, W. (2019). Effects of air injection throat width on a non-premixed rotating detonation engine. *Acta Astronautica*, 159, 189–198.
17. Xia, Z., Ma, H., Liu, C., Zhuo, C., & Zhou, C. (2019). Experimental investigation on the propagation mode of rotating detonation wave in plane-radial combustor. *Experimental Thermal and Fluid Science*, 103, 364–376.
18. Peng, H.-Y., Liu, W.-D., Liu, S.-J., & Zhang, H.-L. (2019). The effect of cavity on ethylene-air Continuous Rotating Detonation in the annular combustor. *International Journal of Hydrogen Energy*, 44(26), 14032–14043.
19. Anand, V., St. George, A., Driscoll, R., & Gutmark, E. (2015). Characterization of instabilities in a Rotating Detonation Combustor. *International Journal of Hydrogen Energy*, 40(46), 16649–16659.
20. Xiao, H., & Oran, E. S. (2019). Shock focusing and detonation initiation at a flame front. *Combustion and Flame*, 203, 397–406.
21. Xia, Z., Tang, X., Luan, M., Zhang, S., Ma, Z., & Wang, J. (2018). Numerical investigation of two-wave collision and wave structure evolution of rotating detonation engine with hollow combustor. *International Journal of Hydrogen Energy*, 43(46), 21582–21591.

22. Tang, X.-M., Wang, J.-P., & Shao, Y.-T. (2015). Three-dimensional numerical investigations of the rotating detonation engine with a hollow combustor. *Combustion and Flame*, 162(4), 997–1008.
23. Katta, V. R., Cho, K. Y., Hoke, J. L., Codoni, J. R., Schauer, F. R., & Roquemore, W. M. (2019). Effect of increasing channel width on the structure of rotating detonation wave. *Proceedings of the Combustion Institute*, 37(3), 3575–3583.
24. Meng, Q., Zhao, N., Zheng, H., Yang, J., & Qi, L. (2018). Numerical investigation of the effect of inlet mass flow rates on H<sub>2</sub>/air non-premixed rotating detonation wave. *International Journal of Hydrogen Energy*, 43(29), 13618–13631.
25. Yao, S., Ma, Z., Zhang, S., Luan, M., & Wang, J. (2017). Reinitiation phenomenon in hydrogen-air rotating detonation engine. *International Journal of Hydrogen Energy*, 42(47), 28588–28598.
26. Yao, S., & Wang, J. (2016). Multiple ignitions and the stability of rotating detonation waves. *Applied Thermal Engineering*, 108, 927–936.
27. Frolov, S. M., Dubrovskii, A. V., & Ivanov, V. S. (2013). Three-dimensional numerical simulation of the operation of a rotating-detonation chamber with separate supply of fuel and oxidizer. *Russian Journal of Physical Chemistry B*, 7(1), 35–43.
28. Liu, M., Zhou, R., & Wang, J.-P. (2014). Numerical Investigation of Different Injection Patterns in Rotating Detonation Engines. *Combustion Science and Technology*, 187(3), 343–361.
29. Yao, S., Tang, X., Luan, M., & Wang, J. (2017). Numerical study of hollow rotating detonation engine with different fuel injection area ratios. *Proceedings of the Combustion Institute*, 36(2), 2649–2655.
30. Tang, X. M., Wang, J. P., & Shao, Y. T. (2015). Three-dimensional numerical investigations of the rotating detonation engine with a hollow combustor. *Combustion and Flame*.
31. Sun, J., Zhou, J., Liu, S., Lin, Z., & Lin, W. (2018). Plume flowfield and propulsive performance analysis of a rotating detonation engine. *Aerospace Science and Technology*, 81, 383–393.

32. Wang, C., Qian, C., Liu, J., & Liberman, M. A. (2018). Influence of chemical kinetics on detonation initiating by temperature gradients in methane/air. *Combustion and Flame*, 197, 400–415.
33. Jin, S., Qi, L., Zhao, N., Zheng, H., Meng, Q., & Yang, J. (2020). Experimental and numerical research on rotating detonation combustor under non-premixed conditions. *International Journal of Hydrogen Energy*, 45(16), 10176–10188.
34. Sun, B., Guo, K., & Pareek, V. K. (2019). Dynamic Simulation on Deflagration of LNG Spill. *Journal of Combustion*, 2019, 1–12.
35. Fan, W., Gao, Y., Zhang, Y., Chow, C., & Chow, W. (2019). Experimental studies and modeling on flame velocity in turbulent deflagration in an open tube. *Process Safety and Environmental Protection*, 129, 291–307.
36. Sutton, I. S. (1992). *Process reliability and risk management*. New York: Van Nostrand Reinhold.
37. Du, W., Ma, Z., Yin, Z., Lv, E., Liu, C., & Hu, E. (2019). Auto-ignition and deflagration characteristics of ethanol-gasoline/air at high temperature. *Fuel*, 255, 115768.
38. Wang, K., Wang, Z., Zhang, Q., Lu, W., Wang, Y., & Fan, W. (2018). Study on a simplified double-frequency scheme for pulse detonation rocket engines. *Acta Astronautica*, 148, 337–344.
39. Zhang, F. (2014). *Shock Waves Science and Technology Library, Vol. 6 Detonation Dynamics*. Berlin: Springer Berlin.
40. Turns, S. R. (2012). *An introduction to combustion: concepts and applications*. New York: McGraw-Hill.
41. Oppenheim, A. K. (2008). *Dynamics of combustion systems*. Berlin: Springer-Verlag.
42. Wolański, P. (2012, November 27). **Detonative propulsion**. Retrieved from <https://www.sciencedirect.com/science/article/pii/S1540748912004014>.
43. Chapman, D. L. (1899). On the rate of explosion in gases. *The London, Edinburgh, and Dublin Philosophical Magazine and Journal of Science*, 47, 90-104.
44. Jouguet, E. (1905). On the propagation of chemical reactions in gases. *J. de mathematiques Pures et Appliquees*, 1, 347-425.
45. Jouguet, E. J. (1906). *Mathem. Pures Appl*, 1, 347-425.
46. **Jouguet, É. (1917). L'œuvre scientifique de Pierre Duhem. Doin.**

47. Zel'dovich; Kompaneets (1960). *Theory of Detonation*. New York: Academic Press.
48. von Neumann, John (1942). Progress report on "Theory of Detonation Waves".
49. Doring, W. (1943). Über den Detonationsvorgang in Gasen. *Annalen der Physik*, 43, 421–436.
50. Erpenbeck, J.J. (1962). Stability of steady-state equilibrium detonations. *Phys. Fluids*, 5, 604–614.
51. Erpenbeck, J.J. (1964). Stability of idealized one-reaction detonations. *Phys. Fluids*, 7, 684–696.
52. Erpenbeck, J.J. (1967). Nonlinear theory of unstable one-dimensional detonations. *Phys. Fluids*, 10, 274–288.
53. Fickett, W. & Wood, W.W. (1966). Flow calculations for pulsating one-dimensional detonations. *Phys. Fluids*, 9, 903–916.
54. Saenz, J. A., Taylor, B. D., & Stewart, D. S. (2012). Asymptotic calculation of the dynamics of self-sustained detonations in condensed phase explosives. *Journal of Fluid Mechanics*, 710, 166–194.
55. Karlin, V. (2010). Radiation Preheating can Trigger Transition from Deflagration to Detonation. *Flow, Turbulence and Combustion*, 87(2-3), 511–523.
56. Toliás, I., Giannissi, S., Venetsanos, A., Keenan, J., Shentsov, V., Makarov, D., ... **Molkov, V. (2019). Best practice guidelines in numerical simulations and CFD benchmarking for hydrogen safety applications. *International Journal of Hydrogen Energy*, 44(17), 9050–9062.**
57. Xiao, J., Breitung, W., Kuznetsov, M., Zhang, H., Travis, J. R., Redlinger, R., & Jordan, T. (2017). GASFLOW-MPI: A new 3-D parallel all-speed CFD code for turbulent dispersion and combustion simulations Part II: First analysis of the hydrogen explosion in Fukushima Daiichi Unit 1. *International Journal of Hydrogen Energy*, 42(12), 8369–8381.
58. Gutiérrez Marcantoni, L., Tamagno, J., & Elaskar, S. (2017). rhoCentralRfFoam: An OpenFOAM solver for high speed chemically active flows – Simulation of planar detonations –. *Computer Physics Communications*, 219, 209–222.

59. Investigation of Deflagration to Detonation Transition for ... (n.d.). Retrieved from  
[http://shepherd.caltech.edu/EDL/publications/reprints/jannaf99\\_paper.pdf](http://shepherd.caltech.edu/EDL/publications/reprints/jannaf99_paper.pdf).
60. Tucker, C., King, P., Bradley, R., & Schauer, F. (2004). The Use of a Flash Vaporization System with Liquid Hydrocarbon Fuels in a Pulse Detonation Engine. *42nd AIAA Aerospace Sciences Meeting and Exhibit*.
61. Michael, & E., J. (1997, September 3). Detonation Database. Retrieved from  
<https://authors.library.caltech.edu/25827/>.
62. Coates, A. M., Mathias, D. L., & Cantwell, B. J. (2019). Numerical investigation of the effect of obstacle shape on deflagration to detonation transition in a hydrogen–air mixture. *Combustion and Flame*, *209*, 278–290.
63. Rakotoarison, W., Maxwell, B., Pekalski, A., & Radulescu, M. I. (2019). Mechanism of flame acceleration and detonation transition from the interaction of a supersonic turbulent flame with an obstruction: Experiments in low pressure propane–oxygen mixtures. *Proceedings of the Combustion Institute*, *37*(3), 3713–3721.
64. Zheng, W., Kaplan, C., Houim, R., & Oran, E. (2019). Flame acceleration and transition to detonation: Effects of a composition gradient in a mixture of methane and air. *Proceedings of the Combustion Institute*, *37*(3), 3521–3528.
65. Xie, Q., Wang, B., Wen, H., He, W., & Wolanski, P. (2019). Enhancement of continuously rotating detonation in hydrogen and oxygen-enriched air. *Proceedings of the Combustion Institute*, *37*(3), 3425–3432.
66. Huang, S., Li, Y., Zhou, J., Liu, S., & Peng, H. (2019). Effects of the pintle injector on H<sub>2</sub>/air continuous rotating detonation wave in a hollow chamber. *International Journal of Hydrogen Energy*, *44*(26), 14044–14054.
67. Kawasaki, A., Inakawa, T., Kasahara, J., Goto, K., Matsuoka, K., Matsuo, A., & Funaki, I. (2019). Critical condition of inner cylinder radius for sustaining rotating detonation waves in rotating detonation engine thruster. *Proceedings of the Combustion Institute*, *37*(3), 3461–3469.
68. Zhou, S., Ma, H., Li, S., Zhou, C., & Liu, D. (2018). Experimental study of a hydrogen-air rotating detonation engine with variable air-inlet slot. *International Journal of Hydrogen Energy*, *43*(24), 11253–11262.
69. Saracoglu, B. H., & Ozden, A. (2018). The effects of multiple detonation waves in the RDE flow field. *Transportation Research Procedia*, *29*, 390–400.



70. Smirnov, N., Nikitin, V., Stamov, L., Mikhalchenko, E., & Tyurenkova, V. (2019). Three-dimensional modeling of rotating detonation in a ramjet engine. *Acta Astronautica*, *163*, 168–176.
71. Olcucuoglu, B., & Saracoglu, B. H. (2018). A preliminary heat transfer analysis of pulse detonation engines. *Transportation Research Procedia*, *29*, 279–288.
72. Sun, J., Zhou, J., Liu, S., & Lin, Z. (2018). Numerical investigation of a rotating detonation engine under premixed/non-premixed conditions. *Acta Astronautica*, *152*, 630–638.
73. Xie, Q., Wen, H., Li, W., Ji, Z., Wang, B., & Wolanski, P. (2018). Analysis of operating diagram for H<sub>2</sub>/Air rotating detonation combustors under lean fuel condition. *Energy*, *151*, 408–419.
74. Zhou, S., Ma, H., Liu, D., Yan, Y., Li, S., & Zhou, C. (2017). Experimental study of a hydrogen-air rotating detonation combustor. *International Journal of Hydrogen Energy*, *42*(21), 14741–14749.
75. Wang, Y., & Wang, J. (2016). Coexistence of detonation with deflagration in rotating detonation engines. *International Journal of Hydrogen Energy*, *41*(32), 14302–14309.
76. Peng, L., Wang, D., Wu, X., Ma, H., & Yang, C. (2015). Ignition experiment with automotive spark on rotating detonation engine. *International Journal of Hydrogen Energy*, *40*(26), 8465–8474.
77. Rankin, B. A., Richardson, D. R., Caswell, A. W., Naples, A., Hoke, J., & Schauer, F. (2015). Imaging of OH\* Chemiluminescence in an Optically Accessible Nonpremixed Rotating Detonation Engine. *53rd AIAA Aerospace Sciences Meeting*.
78. Wang, Y., Wang, J., & Qiao, W. (2016). Effects of thermal wall conditions on rotating detonation. *Computers & Fluids*, *140*, 59–71.
79. Fink, M., Kromer, M., Hillebrandt, W., Röpke, F., Pakmor, R., Seitenzahl, I., & Sim, S. (2018). Thermonuclear explosions of rapidly differentially rotating white dwarfs: Candidates for superluminous Type Ia supernovae? *Astronomy & Astrophysics*, *618*.
80. Mori, K. (2019). Quantum mechanical constraints on resonances in carbon fusion reaction and its impact on type Ia supernovae.

81. Charignon, C., & Chièze, J.-P. (2013). Deflagration-to-detonation transition by amplification of acoustic waves in type Ia supernovae. *Astronomy & Astrophysics*, 550.
82. Leung, S.-C., & Nomoto, K. I. (2018). Explosive Nucleosynthesis in Near-Chandrasekhar-mass White Dwarf Models for Type Ia Supernovae: Dependence on Model Parameters. *The Astrophysical Journal*, 861(2), 143.
83. Ferrand, G., Warren, D. C., Ono, M., Nagataki, S., Röpke, F. K., & Seitenzahl, I. R. (2019). From Supernova to Supernova Remnant: The Three-dimensional Imprint of a Thermonuclear Explosion. *The Astrophysical Journal*, 877(2), 136.
84. Thomas, G. O., Bambrey, R. J., & Oakley, G. L. (2018). A study of flame acceleration and the possibility of detonation with silane mixtures. *Process Safety and Environmental Protection*, 117, 278–285.
85. Toliás, I., & Venetsanos, A. (2018). An improved CFD model for vented deflagration simulations – Analysis of a medium-scale hydrogen experiment. *International Journal of Hydrogen Energy*, 43(52), 23568–23584.
86. Xu, H., Gao, J., Yao, A., & Yao, C. (2018). The effect of the energy convergence and energy dissipation on the formation of severe knock. *Applied Energy*, 228, 1243–1254.
87. Kim, N. K., Jeon, J., Choi, W., & Kim, S. J. (2018). Systematic hydrogen risk analysis of OPR1000 containment before RPV failure under station blackout scenario. *Annals of Nuclear Energy*, 116, 429–438.
88. **Wang, B., Rao, Z., Xie, Q., Wolański, P., & Rarata, G. (2017). Brief review on passive and active methods for explosion and detonation suppression in tubes and galleries. *Journal of Loss Prevention in the Process Industries*, 49, 280–290.**
89. Davis, S., Merilo, E., Engel, D., Ziemba, A., Pinto, M., & Wingerden, K. V. (2017). Large scale detonation testing: New findings in the prediction of DDTs at large scales. *Journal of Loss Prevention in the Process Industries*, 48, 345–357.
90. Klebanoff, L., Pratt, J., & Lafleur, C. (2017). Comparison of the safety-related physical and combustion properties of liquid hydrogen and liquid natural gas in the context of the SF-BREEZE high-speed fuel-cell ferry. *International Journal of Hydrogen Energy*, 42(1), 757–774.

91. Hasslberger, J., Kim, H. K., Kim, B. J., Ryu, I. C., & Sattelmayer, T. (2017). Three-dimensional CFD analysis of hydrogen-air-steam explosions in APR1400 containment. *Nuclear Engineering and Design*, 320, 386–399.
92. Baraldi, D., Melideo, D., Kotchourko, A., Ren, K., Yanez, J., Jedicke, O., ... Duclos, A. (2017). Development of a model evaluation protocol for CFD analysis of hydrogen safety issues the SUSANA project. *International Journal of Hydrogen Energy*, 42(11), 7633–7643.
93. Wang, C., Zhao, Y., & Zhang, B. (2016). Numerical simulation of flame acceleration and deflagration-to-detonation transition of ethylene in channels. *Journal of Loss Prevention in the Process Industries*, 43, 120–126.
94. Yang, T., Escobar, S., & Celik, I. B. (2016). Understanding flame trapping in detonation combustion via vacuum chambers. *Journal of Loss Prevention in the Process Industries*, 41, 231–240.
95. Zhang, B., Xiu, G., Chen, J., & Yang, S. (2015). Detonation and deflagration characteristics of p-Xylene/gaseous hydrocarbon fuels/air mixtures. *Fuel*, 140, 73–80.
96. Boeck, L., Kink, A., Oezdin, D., Hasslberger, J., & Sattelmayer, T. (2015). Influence of water mist on flame acceleration, DDT and detonation in H<sub>2</sub>-air mixtures. *International Journal of Hydrogen Energy*, 40(21), 6995–7004.
97. Pekalski, A., Puttock, J., & Chynoweth, S. (2015). Deflagration to detonation transition in a vapour cloud explosion in open but congested space: Large scale test. *Journal of Loss Prevention in the Process Industries*, 36, 365–370.
98. Wei, H., Shang, Y., Chen, C., Gao, D., & Feng, D. (2015). One-dimensional numerical study on pressure wave–flame interaction and flame acceleration under engine-relevant conditions. *International Journal of Hydrogen Energy*, 40(14), 4874–4883.
99. George, A. C. S., Driscoll, R. B., Munday, D. E., & Gutmark, E. J. (2015). Development of a Rotating Detonation Engine Facility at the University of Cincinnati. *53rd AIAA Aerospace Sciences Meeting*.
100. Heister, S. D., & Stechmann, D. P. (2014). Survey of Rotating Detonation Wave Combustor Technology and Potential Rocket Vehicle Applications. *50th AIAA/ASME/SAE/ASEE Joint Propulsion Conference*.
101. Voitsekhovskii, B.V. (1960) Stationary spin detonation. *Sov J Appl Mech Tech Phys*.

102. Adamson, T. C. J. & Olsson, G. R. (1967). Performance analysis of a rotating detonation wave rocket engine. *Astronaut Acta*.
103. Nicholls, J. A., Cullen, R. E. & Ragland, K. W. (1966). Feasibility studies of a rotating detonation wave rocket motor. *J Spacecraft Rockets*.
104. Bykovskii, F. A., Zhdan, S. A. & Evgenii FV. (2006). Continuous spin detonations. *J Propul Power*.
105. Xie, Q., Wang, B., Wen, H., He, W., & Wolanski, P. (2019). Enhancement of continuously rotating detonation in hydrogen and oxygen-enriched air. *Proceedings of the Combustion Institute*, 37(3), 3425–3432.
106. Burr, J. R., & Yu, K. H. (2019). Experimental characterization of RDE combustor flowfield using linear channel. *Proceedings of the Combustion Institute*, 37(3), 3471–3478.
107. Zhou, S., Ma, H., Yang, Y., & Zhou, C. (2019). Investigation on propagation characteristics of rotating detonation wave in a radial-flow turbine engine combustor model. *Acta Astronautica*, 160, 15–24.
108. Wen, H., Xie, Q., & Wang, B. (2019). Propagation behaviours of rotating detonation in an obround combustor. *Combustion and Flame*, 210, 389–398.
109. Smirnov, N., Nikitin, V., Stamov, L., Mikhilchenko, E., & Tyurenkova, V. (2019). Three-dimensional modeling of rotating detonation in a ramjet engine. *Acta Astronautica*, 163, 168–176.
110. Ma, J. Z., Zhang, S., Luan, M., & Wang, J. (2019). Experimental investigation on delay time phenomenon in rotating detonation engine. *Aerospace Science and Technology*, 88, 395–404.
111. Zhong, Y., Wu, Y., Jin, D., Chen, X., Yang, X., & Wang, S. (2019). Effect of channel and oxidizer injection slot width on the rotating detonation fueled by pre-combustion cracked kerosene. *Acta Astronautica*, 165, 365–372.
112. Ge, G., Deng, L., Ma, H., Liu, X., Jin, L., & Zhou, C. (2019). Effect of blockage ratio on the existence of multiple waves in rotating detonation engine. *Acta Astronautica*, 164, 230–240.
113. Rankin, B. A., Codoni, J. R., Cho, K. Y., Hoke, J. L., & Schauer, F. R. (2019). Investigation of the structure of detonation waves in a non-premixed hydrogen–air rotating detonation engine using mid-infrared imaging. *Proceedings of the Combustion Institute*, 37(3), 3479–3486.

114. Wang, Y., & Le, J. (2019). A hollow combustor that intensifies rotating detonation. *Aerospace Science and Technology*, *85*, 113–124.
115. Peng, H.-Y., Liu, W.-D., Liu, S.-J., Zhang, H.-L., & Zhou, W.-Y. (2019). Realization of methane-air continuous rotating detonation wave. *Acta Astronautica*, *164*, 1–8.
116. Li, B., Wu, Y., Weng, C., Zheng, Q., & Wei, W. (2018). Influence of equivalence ratio on the propagation characteristics of rotating detonation wave. *Experimental Thermal and Fluid Science*.
117. Zhong, Y., Jin, D., Wu, Y., & Chen, X. (2018). Investigation of rotating **detonation wave fueled by “ethylene-acetylene-hydrogen” mixture**. *International Journal of Hydrogen Energy*, *43*(31), 14787–14797.
118. Xia, Z., Ma, H., Zhuo, C., & Zhou, C. (2018). Propagation process of H<sub>2</sub>/air rotating detonation wave and influence factors in plane-radial structure. *International Journal of Hydrogen Energy*, *43*(9), 4609–4622.
119. Li, Q., Liu, P., & Zhang, H. (2018). Further investigations on the interface instability between fresh injections and burnt products in 2-D rotating detonation. *Computers & Fluids*, *170*, 261–272.
120. XUE, S., LIU, H., ZHOU, L., YANG, W., HU, H., & YAN, Y. (2018). Experimental research on rotating detonation with liquid hypergolic propellants. *Chinese Journal of Aeronautics*, *31*(12), 2199–2205.
121. Anand, V., St. George, A., Farbos de Luzan, C., & Gutmark, E. (2018). Rotating detonation wave mechanics through ethylene-air mixtures in hollow combustors, and implications to high frequency combustion instabilities. *Experimental Thermal and Fluid Science*, *92*, 314–325.
122. Deng, L., Ma, H., Xu, C., Liu, X., & Zhou, C. (2018). The feasibility of mode control in rotating detonation engine. *Applied Thermal Engineering*, *129*, 1538–1550.
123. Zhang, H., Liu, W., & Liu, S. (2017). Experimental investigations on H<sub>2</sub>/air rotating detonation wave in the hollow chamber with Laval nozzle. *International Journal of Hydrogen Energy*, *42*(5), 3363–3370.
124. Gaillard, T., Davidenko, D., & Dupoirieux, F. (2017). Numerical simulation of a Rotating Detonation with a realistic injector designed for separate supply of gaseous hydrogen and oxygen. *Acta Astronautica*, *141*, 64–78.

125. Frolov, S. M., Aksenov, V. S., Ivanov, V. S., & Shamshin, I. O. (2017). **Continuous detonation combustion of ternary “hydrogen–liquid propane–air” mixture in annular combustor.** *International Journal of Hydrogen Energy*, *42*(26), 16808–16820.
126. Sun, J., Zhou, J., Liu, S., Lin, Z., & Cai, J. (2017). Effects of injection nozzle exit width on rotating detonation engine. *Acta Astronautica*, *140*, 388–401.
127. Nakagami, S., Matsuoka, K., Kasahara, J., Matsuo, A., & Funaki, I. (2017). Experimental study of the structure of forward-tilting rotating detonation waves and highly maintained combustion chamber pressure in a disk-shaped combustor. *Proceedings of the Combustion Institute*, *36*(2), 2673–2680.
128. Deng, L., Ma, H., Xu, C., Zhou, C., & Liu, X. (2017). Investigation on the propagation process of rotating detonation wave. *Acta Astronautica*, *139*, 278–287.
129. Anand, V., St. George, A., & Gutmark, E. (2017). Amplitude modulated instability in reactants plenum of a rotating detonation combustor. *International Journal of Hydrogen Energy*, *42*(17), 12629–12644.
130. Anand, V., St. George, A., Driscoll, R., & Gutmark, E. (2016). Longitudinal pulsed detonation instability in a rotating detonation combustor. *Experimental Thermal and Fluid Science*, *77*, 212–225.
131. Driscoll, R., St. George, A., & Gutmark, E. J. (2016). Numerical investigation of injection within an axisymmetric rotating detonation engine. *International Journal of Hydrogen Energy*, *41*(3), 2052–2063.
132. Zhang, H., Liu, W., & Liu, S. (2016). Effects of inner cylinder length on H<sub>2</sub>/air rotating detonation. *International Journal of Hydrogen Energy*, *41*(30), 13281–13293.
133. Driscoll, R., Aghasi, P., St George, A., & Gutmark, E. J. (2016). Three-dimensional, numerical investigation of reactant injection variation in a H<sub>2</sub>/air rotating detonation engine. *International Journal of Hydrogen Energy*.
134. Kindracki, J. (2015). Experimental research on rotating detonation in liquid fuel–gaseous air mixtures. *Aerospace Science and Technology*, *43*, 445–453.
135. Lin, W., Zhou, J., Liu, S., & Lin, Z. (2015). An experimental study on CH<sub>4</sub>/O<sub>2</sub> continuously rotating detonation wave in a hollow combustion chamber. *Experimental Thermal and Fluid Science*, *62*, 122–130.

136. Lin, W., Zhou, J., Liu, S., Lin, Z., & Zhuang, F. (2015). Experimental study on propagation mode of H<sub>2</sub>/Air continuously rotating detonation wave. *International Journal of Hydrogen Energy*, 40(4), 1980–1993.
137. Wang, C., Liu, W., Liu, S., Jiang, L., & Lin, Z. (2015). Experimental investigation on detonation combustion patterns of hydrogen/vitiated air within annular combustor. *Experimental Thermal and Fluid Science*, 66, 269–278.
138. Wang, Y., & Wang, J. (2015). Effect of equivalence ratio on the velocity of rotating detonation. *International Journal of Hydrogen Energy*, 40(25), 7949–7955.
139. Yu-Si, L., Yang, L., Yu-hui, W., & Jian-ping, W. (2015). Research on the Influence of Reflected Shock Wave on Continuously Rotating Detonation Engine. *Procedia Engineering*, 99, 1263–1267.
140. Sun, J., Zhou, J., Liu, S., & Lin, Z. (2019). Interaction between rotating detonation wave propagation and reactant mixing. *Acta Astronautica*, 164, 197–203.
141. Kawasaki, A., Inakawa, T., Kasahara, J., Goto, K., Matsuoka, K., Matsuo, A., & Funaki, I. (2019). Critical condition of inner cylinder radius for sustaining rotating detonation waves in rotating detonation engine thruster. *Proceedings of the Combustion Institute*, 37(3), 3461–3469.
142. Ji, Z., Zhang, H., & Wang, B. (2019). Performance analysis of dual-duct rotating detonation aero-turbine engine. *Aerospace Science and Technology*, 92, 806–819.
143. Tsuboi, N., Watanabe, Y., Kojima, T., & Hayashi, A. K. (2015). Numerical estimation of the thrust performance on a rotating detonation engine for a hydrogen–oxygen mixture. *Proceedings of the Combustion Institute*, 35(2), 2005–2013.
144. Jourdaine, N., Tsuboi, N., Ozawa, K., Kojima, T., & Koichi Hayashi, A. (2018). Three-dimensional numerical thrust performance analysis of hydrogen fuel mixture rotating detonation engine with aerospike nozzle. *Proceedings of the Combustion Institute*, 37(3), 3443–3451.
145. Yao, S. B., Liu, M., & Wang, J. P. (2015). The Effect of the Inlet Total Pressure and the Number of Detonation Waves on Rotating Detonation Engines. *Procedia Engineering*.

146. Yang, X., Song, F., Wu, Y., Zhong, Y., & Xu, S. (2020). Investigation of rotating detonation fueled by a methane–hydrogen–carbon dioxide mixture under lean fuel conditions. *International Journal of Hydrogen Energy*, *45*(41), 21995-22007.
147. Xia, Z., Luan, M., Liu, X., & Wang, J. (2020). Numerical simulation of wave mode transition in rotating detonation engine with OpenFOAM. *International Journal of Hydrogen Energy*, *45*(38), 19989-19995.
148. Sato, T., Chacon, F., White, L., Raman, V., & Gamba, M. (2021). Mixing and detonation structure in a rotating detonation engine with an axial air inlet. *Proceedings of the Combustion Institute*, *38*(3), 3769-3776.
149. Zhao, M., & Zhang, H. (2020). Large eddy simulation of non-reacting flow and mixing fields in a rotating detonation engine. *Fuel*, *280*, 118534.
150. Yokoo, R., Goto, K., Kasahara, J., Athmanathan, V., Braun, J., Paniagua, G., . . . Funaki, I. (2021). Experimental study of internal flow structures in cylindrical rotating detonation engines. *Proceedings of the Combustion Institute*, *38*(3), 3759-3768.
151. Wu, K., Zhang, S., Luan, M., & Wang, J. (2020). Effects of flow-field structures on the stability of rotating detonation ramjet engine. *Acta Astronautica*, *168*, 174-181.
152. Liu, X., Luan, M., Chen, Y., & Wang, J. (2020). Flow-field analysis and pressure gain estimation of a rotating detonation engine with banded distribution of reactants. *International Journal of Hydrogen Energy*, *45*(38), 19976-19988.
153. Liu, X., Chen, Y., Xia, Z., & Wang, J. (2020). Numerical study of the reverse-rotating waves in rotating detonation engine with a hollow combustor. *Acta Astronautica*, *170*, 421-430.
154. Wang, F., Weng, C., Wu, Y., Bai, Q., Zheng, Q., & Xu, H. (2020). Numerical research on kerosene/air rotating detonation engines under different injection total temperatures. *Aerospace Science and Technology*, *103*, 105899.
155. Sosa, J., Ahmed, K. A., Fievisohn, R., Hoke, J., Ombrello, T., & Schauer, F. (2019). Supersonic driven detonation dynamics for rotating detonation engines. *International Journal of Hydrogen Energy*, *44*(14), 7596-7606.
156. Zheng, H., Meng, Q., Zhao, N., Li, Z., & Deng, F. (2020). Numerical investigation on H<sub>2</sub>/air non-premixed rotating detonation engine under



- different equivalence ratios. *International Journal of Hydrogen Energy*, 45(3), 2289-2307.
157. Liu, P., Guo, Q., Sun, D., Li, C., & Zhang, H. (2020). Wall effect on the flow structures of Three-dimensional rotating detonation wave. *International Journal of Hydrogen Energy*, 45(53), 29546-29559.
  158. Zhao, M., & Zhang, H. (2020). Origin and chaotic propagation of multiple rotating detonation waves in hydrogen/air mixtures. *Fuel*, 275, 117986.
  159. Wei, W., Wu, Y., Weng, C., & Zheng, Q. (2020). Influence of propagation direction on operation performance of rotating detonation combustor with turbine guide vane. *Defence Technology*.
  160. Xia, Z., Ma, H., Ge, G., & Zhou, C. (2020). Visual experimental investigation On initiation process of H<sub>2</sub>/air rotating detonation wave in plane-radial structure. *International Journal of Hydrogen Energy*, 45(53), 29579-29593.
  161. Wang, Z., Wang, K., Li, Q., Zhu, Y., Zhao, M., & Fan, W. (2020). Effects of the combustor width on propagation characteristics of rotating detonation waves. *Aerospace Science and Technology*, 105, 106038.
  162. Zheng, Q., Meng, H., Weng, C., Wu, Y., Feng, W., & Wu, M. (2020). Experimental research on the instability propagation characteristics of liquid kerosene rotating detonation wave. *Defence Technology*, 16(6), 1106-1115.
  163. Peng, H., Liu, W., Liu, S., Zhang, H., & Jiang, L. (2020). Hydrogen-air, ethylene-air, and methane-air continuous rotating detonation in the hollow chamber. *Energy*, 211, 118598.
  164. Wen, H., & Wang, B. (2020). Experimental study of perforated-wall rotating detonation combustors. *Combustion and Flame*, 213, 52-62.
  165. Teng, H., Zhou, L., Yang, P., & Jiang, Z. (2020). Numerical investigation of wavelet features in rotating detonations with a two-step induction-reaction model. *International Journal of Hydrogen Energy*, 45(7), 4991-5001.
  166. Wang, Y., Huang, C., Deiterding, R., Chen, H., & Chen, Z. (2021). Propagation of gaseous detonation across inert layers. *Proceedings of the Combustion Institute*, 38(3), 3555-3563.
  167. Liu, S., Peng, H., Liu, W., & Zhang, H. (2020). Effects of cavity depth on the ethylene-air continuous rotating detonation. *Acta Astronautica*, 166, 1-10.

168. Betelin, V., Nikitin, V., & Mikhilchenko, E. (2020). 3D numerical modeling of a cylindrical rde with an inner body extending out of the nozzle. *Acta Astronautica*, 176, 628-646.
169. Zhou, S., Ma, H., Li, S., Liu, D., Yan, Y., & Zhou, C. (2017). Effects of a turbine guide vane on hydrogen-air rotating detonation wave propagation characteristics. *International Journal of Hydrogen Energy*, 42(31), 20297-20305.
170. Wang, Y., & Le, J. (2021). A rotating detonation engine using methane-ethylene mixture and air. *Acta Astronautica*, 188, 25–35.
171. Yan, C., Teng, H., & Ng, H. D. (2021). Effects of slot injection on detonation wavelet characteristics in a rotating detonation engine. *Acta Astronautica*, 182, 274–285.
172. Dunn, I. B., Malik, V., Flores, W., Morales, A., & Ahmed, K. A. (2021). Experimental and theoretical analysis of carbon driven detonation waves in a heterogeneously premixed rotating detonation engine. *Fuel*, 302, 121128.
173. Zhou, S., Ma, H., Ma, Y., Zhou, C., & Hu, N. (2021). Experimental investigation on detonation wave propagation mode in the start-up process of rotating detonation turbine engine. *Aerospace Science and Technology*, 111, 106559.
174. Walters, I. V., Gejji, R. M., Heister, S. D., & Slabaugh, C. D. (2021). Flow and performance analysis of a natural gas-air rotating detonation engine with high-speed velocimetry. *Combustion and Flame*, 232, 111549.
175. Nair, A. P., Lee, D. D., Pineda, D. I., Kriesel, J., Hargus, W. A., Bennowitz, J. W., Bigler, B., Danczyk, S. A., & Spearrin, R. M. (2021). Methane-oxygen rotating detonation exhaust thermodynamics with variable mixing, equivalence ratio, and mass flux. *Aerospace Science and Technology*, 113, 106683.
176. Ren, Z., & Zheng, L. (2021). Numerical study on rotating detonation stability in Two-phase kerosene-air mixture. *Combustion and Flame*, 231, 111484.
177. Liu, X.-Y., Luan, M.-Y., Chen, Y.-L., & Wang, J.-P. (2021). Propagation behavior of rotating detonation waves with premixed kerosene/air mixtures. *Fuel*, 294, 120253.
178. Meng, H., Zheng, Q., Weng, C., Wu, Y., Feng, W., Xu, G., & Wang, F. (2021). Propagation mode analysis of rotating detonation waves fueled by liquid kerosene. *Acta Astronautica*, 187, 248-258.

179. Zhang, H., Liu, W., & Liu, S. (2021). Research on H<sub>2</sub>/air rotating detonation in the hollow chamber with double injection. *International Journal of Hydrogen Energy*, 46(44), 23067–23074.
180. Huang, Y., Xia, H., Chen, X., Luan, Z., & You, Y. (2021). Shock dynamics and expansion characteristics of an aerospike nozzle and its interaction with the rotating detonation combustor. *Aerospace Science and Technology*, 117, 106969.
181. Burke, R., Rezzag, T., Dunn, I., Flores, W., & Ahmed, K. (2021). The effect of premixed stratification on the wave dynamics of a rotating detonation combustor. *International Journal of Hydrogen Energy*, 46(54), 27816–27826.
182. Sies, M. M. (2021). Performance and Characterization of Asymmetric Vortex Combustor Using Different Compositions of Synthetic Biogas (Unpublished PHD's thesis). Universiti Teknologi Malaysia, Johor, Malaysia.
183. Lu, F. K., & Braun, E. M. (2014). Rotating Detonation Wave Propulsion: Experimental Challenges, Modeling, and Engine Concepts. *Journal of Propulsion and Power*, 30(5), 1125–1142.
184. St. George, A., Randall, S., Anand, V., Driscoll, R., & Gutmark, E. (2016). Characterization of initiator dynamics in a rotating detonation combustor. *Experimental Thermal and Fluid Science*, 72, 171–181.
185. Saqr, K. M., Kassem, H. I., Mohsin, M. S., & Mazlan, A. W. (2010). Ideal detonation characteristics of biogas-hydrogen and – hydrogen peroxide mixtures. *Latest Trends on Theoretical and Applied Mechanics, Fluid Mechanics and Heat & Mass Transfer*
186. Zheng, L., Dou, Z., Du, D., Wang, X., Jin, H., Yu, M., & Wang, Y. (2019). Study on explosion characteristics of premixed hydrogen/biogas/air mixture in a duct. *International Journal of Hydrogen Energy*, 44(49), 27159–27173.
187. Wei, S., Yu, M., Pei, B., Zhu, Z., & Zhang, Z. (2020). Suppression of CO<sub>2</sub> and H<sub>2</sub>O on the cellular instability of premixed methane/air flame. *Fuel*, 264, 116862.
188. Sulaiman, S. Z., Khan, N. A., Izhah, I., Md. Shaarani, S., Mudalip, S. K., Man, R. C., Arshad, Z. I., Kasmani, R. M., & Sulaiman, S. (2020). Explosion characteristics assessment of premixed biogas/air mixture in a 20-L spherical vessel. *Chemical Engineering Communications*, 208(4), 583–591.

189. Khan, N. A., Sulaiman, S. Z., Izhab, I., Mudalip, S. K., Che Man, R., Md Shaarani, S., Mohd Arshad, Z. I., Kasmani, R. M., & Sulaiman, S. (2019). The Explosion Severity of Biogas(CH<sub>4</sub>-CO<sub>2</sub>)/Air Mixtures in a Closed Vessel. *Materials Science Forum*, *964*, 33–39.
190. Zheng, K., Yang, X., Yu, M., Si, R., & Wang, L. (2019). Effect of N<sub>2</sub> and CO<sub>2</sub> on explosion behavior of syngas/air mixtures in a closed duct. *International Journal of Hydrogen Energy*, *44*(51), 28044–28055.
191. Shen, X., Zhang, N., Shi, X., & Cheng, X. (2019). Experimental studies on pressure dynamics of C<sub>2</sub>H<sub>4</sub>/N<sub>2</sub>O mixtures explosion with dilution. *Applied Thermal Engineering*, *147*, 74–80.
192. Yuen, K. K., & Yeoh, G. H. (2010). *Computational fluid dynamics in fire engineering: theory, modelling and practice*. Amsterdam.
193. Anderson, J. D. (2010). *Computational fluid dynamics: The basics with applications*. New York, NY: McGraw-Hill.
194. Kundu, P. K., Cohen, I. M., & Dowling, D. R. (2016). *Fluid mechanics*. Amsterdam: Elsevier.
195. Dounia, O., Vermorel, O., Misdariis, A., & Poinso, T. (2019). Influence of kinetics on DDT simulations. *Combustion and Flame*, *200*, 1–14.
196. Emami, S., Mazaheri, K., Shamooni, A., & Mahmoudi, Y. (2015). LES of flame acceleration and DDT in hydrogen–air mixture using artificially thickened flame approach and detailed chemical kinetics. *International Journal of Hydrogen Energy*, *40*(23), 7395–7408.
197. Zhao, Y., Wang, C., & Bi, Y. (2017). LES of flame acceleration and DDT in small-scale channels. *Journal of Loss Prevention in the Process Industries*, *49*, 745–752.
198. Malik, K., Żbikowski, M., Bąk, D., Lesiak, P., & Teodorczyk, A. (2019). Numerical and experimental investigation of H<sub>2</sub>-air and H<sub>2</sub>O<sub>2</sub> detonation parameters in a 9 m long tube, introduction of a new detonation model. *International Journal of Hydrogen Energy*, *44*(17), 8743–8750.
199. Zhou, F., Liu, N., & Zhang, X. (2018). Numerical study of hydrogen–oxygen flame acceleration and deflagration to detonation transition in combustion light gas gun. *International Journal of Hydrogen Energy*, *43*(10), 5405–5414.
200. Azadboni, R. K., Heidari, A., Boeck, L. R., & Wen, J. X. (2019). The effect of concentration gradients on deflagration-to-detonation transition in a

- rectangular channel with and without obstructions – A numerical study. *International Journal of Hydrogen Energy*, 44(13), 7032-7040.
201. Maxwell, B., Pekalski, A., & Radulescu, M. (2018). Modelling of the transition of a turbulent shock-flame complex to detonation using the linear eddy model. *Combustion and Flame*, 192, 340-357.
202. Zheng, Y., Wang, C., Wang, Y., Liu, Y., & Yan, Z. (2019). Numerical research of rotating detonation initiation processes with different injection patterns. *International Journal of Hydrogen Energy*, 44(29), 15536-15552.
203. Wang, Y. (2016). Rotating detonation in a combustor of trapezoidal cross section for the hydrogen–air mixture. *International Journal of Hydrogen Energy*, 41(12), 5605-5616.
204. Cai, X., Liang, J., Deiterding, R., & Lin, Z. (2016). Adaptive simulations of cavity-based detonation in supersonic hydrogen–oxygen mixture. *International Journal of Hydrogen Energy*, 41(16), 6917-6928.
205. Chen, W., Liang, J., Cai, X., & Lin, Z. (2017). Detonation simulations in expanding channel with supersonic combustible mixture. *International Journal of Hydrogen Energy*, 42(9), 6384-6393.
206. Kiverin, A., & Yakovenko, I. (2019). Ignition and detonation onset behind incident shock wave in the shock tube. *Combustion and Flame*, 204, 227-236.
207. Wang, T., Zhang, Y., Teng, H., & Jiang, Z. (2015). Oblique Shock to Detonation Transition in Hydrogen-air Mixtures. *Procedia Engineering*, 126, 209-213.
208. Azadboni, R. K., Wen, J. X., Heidari, A., & Wang, C. (2017). Numerical modeling of deflagration to detonation transition in inhomogeneous hydrogen/air mixtures. *Journal of Loss Prevention in the Process Industries*, 49, 722-730.
209. Karanam, A., Sharma, P. K., & Ganju, S. (2018). Numerical simulation and validation of flame acceleration and DDT in hydrogen air mixtures. *International Journal of Hydrogen Energy*, 43(36), 17492-17504.
210. Wang, C., & Wen, J. (2017). Numerical simulation of flame acceleration and deflagration-to-detonation transition in hydrogen-air mixtures with concentration gradients. *International Journal of Hydrogen Energy*, 42(11), 7657-7663.

211. Luo, C., Zanganeh, J., & Moghtaderi, B. (2016). A 3D numerical study on the effects of obstacles on flame propagation in a cylindrical explosion vessel connected to a vented tube. *Journal of Loss Prevention in the Process Industries*, 44, 53–61.
212. ANSYS FLUENT Theory Guide 2019 R1. Retrieved from <https://studentcommunity.ansys.com/thread/theory-guide-2019r1/>.
213. Oran, E. S., & Gamezo, V. N. (2007). Origins of the deflagration-to-detonation transition in gas-phase combustion. *Combustion and Flame*, 148(1-2), 4–47.
214. Zhang, B., Liu, H., & Wang, C. (2017). On the detonation propagation behaviour in hydrogen-oxygen mixture under the effect of spiral obstacles. *International Journal of Hydrogen Energy*, 42(33), 21392–21402.
215. Goodwin, G. B., Houim, R. W., & Oran, E. S. (2017). Shock transition to detonation in channels with obstacles. *Proceedings of the Combustion Institute*, 36(2), 2717–2724.
216. Heidari, A., & Mazaheri, K. (2008). Determination of the Equilibrium Parameters of Gaseous Detonations Using a Genetic Algorithm. *Iranian Journal of Chemical Engineering*, 5(3).
217. Gordon, S., & McBride, B. J. (1994). Computer Program for Calculation Complex Chemical Equilibrium Compositions and Applications (I. Analysis). *NASA Reference Publication 1311*.
218. McBride, B. J., & Gordon, S. (1996). Computer Program for Calculation Complex Chemical Equilibrium Compositions and Applications (II. Users Manual and Program Description). *NASA Reference Publication 1311*.
219. Menter, F. R. (1994). Two-equation eddy-viscosity turbulence models for engineering applications. *AIAA Journal*, 32(8), 1598–1605.
220. Menter, F. R. (2009). Review of the shear-stress transport turbulence model experience from an industrial perspective. *International Journal of Computational Fluid Dynamics*, 23(4), 305–316.
221. Westbrook, C. K., & Dryer, F. L. (1981). Simplified Reaction Mechanisms for the Oxidation of Hydrocarbon Fuels in Flames. *Combustion Science and Technology*, 27(1-2), 31–43.
222. Silva, V. B., & João, C. (2020). *Computational fluid dynamics applied to waste-to-energy processes: a hands-on approach*. Butterworth-Heinemann.

223. Acampora, L., Marra, F. S., & Martelli, E. (2016). Comparison of Different CH<sub>4</sub>-air Combustion mechanisms in a Perfectly Stirred reactor with Oscillating residence times close to extinction. *Combustion Science and Technology*, 188(4-5), 707–718.
224. Lee, J. H. S. (2008). *The Detonation Phenomenon*. Cambridge University Press, New York
225. Theuerkauf, S. W., Schauer, F., Anthony, R., & Hoke, J. (2015). Experimental Characterization of High-Frequency Heat Flux in a Rotating Detonation Engine. 53rd AIAA Aerospace Sciences Meeting.
226. Rahman, M. N., & Wahid, M. A. (2021). Renewable-based zero-carbon fuels for the use of power generation: A case study in Malaysia supported by updated developments worldwide. *Energy Report*, 7, 1986–2020.

## LIST OF PUBLICATIONS

1. Rahman, M. and Wahid, M., 2021. Renewable-based zero-carbon fuels for the use of power generation: A case study in Malaysia supported by updated developments worldwide. *Energy Reports*, 7, pp.1986-2020.
2. Rahman, M., Wahid, M., Yasin, M., Abidin, A. and Mazlan, A., 2021. Predictive Numerical Analysis on the Mixing Characteristics in a Rotating Detonation Engine (RDE). *Evergreen*, 8(1), pp.123-130.
3. Mazlan, M., Yasin, M., Saat, A., Wahid, M., Ghazali, A. and Rahman, M., 2021. Initiation Characteristics of Rotating Supersonic Combustion Engine. *Evergreen*, 8(1), pp.177-181.
4. Rahman, M., Wahid, M. and Yasin, M., 2020. Predictive Numerical Analysis on the Fuel Homogeneity in a Rotating Detonation Engine (RDE) Implementing Radially-Entered Fuel Injection Scheme. *IOP Conference Series: Materials Science and Engineering* 884 (1), 012109
5. Rahman, M., Wahid, M. and Yasin, M., Detonation propulsion longevity in biogas- and hydrogen-fuelled rotating detonation engines. *Aerospace Science and Technology* (under review)
6. Rahman, M., Ujir, M., Wahid, M. and Yasin, M., A Single-Step Chemistry Mechanism for Biogas Supersonic Combustion Velocity with Nitrogen Dilution. *Journal of Thermal Analysis and Calorimetry* (under review)